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PLANE STRAIN FINITE ELEMENT PROGRAM
LAMINATED COMPOSITES SUBJECT TO MECHANICAL
AND HYGROTHERMAL LOADING Interim Report
(Virginia Polytechnic Inst. and State Univ.)

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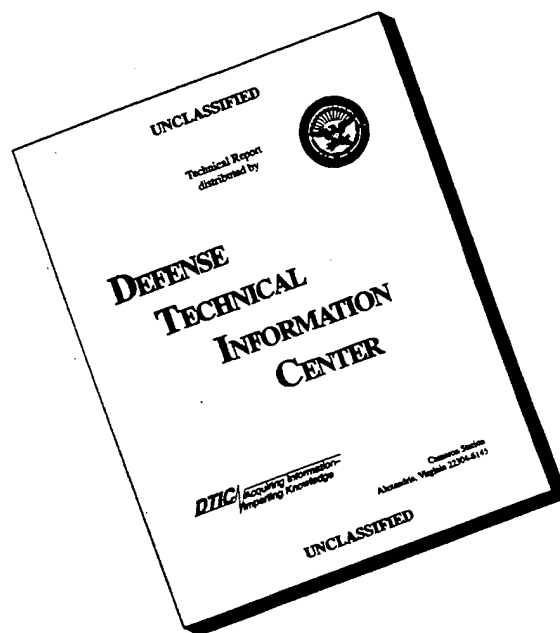
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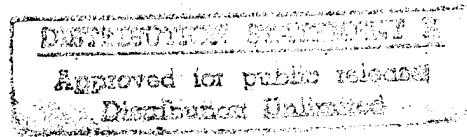
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College of Engineering
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Blacksburg, VA 24061



VPI-E-83-40

October, 1983

CLFE2D - A Generalized Plane Strain Finite
Element Program for Laminated Composites Subjected
to Mechanical and Hygrothermal Loading

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16. Abstracts			
<p>CLFE2D is a two dimensional generalized plane strain finite element code. It uses a linear, four node, general quadrilateral, isoparametric element.</p> <p>The program was developed to calculate the displacements, strains, stresses, and strain energy densities in a finite width composite laminate. CLFE2D offers any combination of the following load types: nodal displacements, nodal forces, uniform normal strain, or hygrothermal.</p> <p>The program allows the user to input one set of three dimensional orthotropic material properties. The user can then specify the angle of material principal orientation for each element in the mesh.</p> <p>Output includes displacements, stresses, strains and strain energy densities at points selected by the user. An option is also available to plot the undeformed and deformed finite element meshes.</p>			
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1.0 Introduction:

CLFE2D was developed under the assumption of a state of generalized plane strain. The most general displacement field for such problems can be written

$$u_{total} = U(y,z) + \epsilon_x^0 x \quad (1a)$$

$$v = V(y,z) \quad (1b)$$

$$w = W(y,z) \quad (1c)$$

The assumption of generalized plane strain allows the user to study the response of an infinitely long body of arbitrary cross-section to various loadings with the restriction that all quantities except axial displacement are independent of the axial coordinate x . The coordinate system used by CLFE2D is shown in Fig. (1.1). The x , y and z axes represent the global coordinate system while the material principal directions are represented by the 1, 2 and 3 axes. The region of interest to be modeled by finite elements is then a typical cross section (or a symmetric portion of it).

The element used in CLFE2D is a four node, linear, general quadrilateral, isoparametric element. A three node version of the element can be obtained by "double-numbering" the first local node in the element connectivity data. The available element geometries are shown in Fig. (1.2). CLFE2D requires that the local node numbering scheme, for a given element, proceed in a counter-clockwise direction.

An in-depth description of the finite element used in this program can be found in reference [1]. The finite element formulation for the generalized plane strain problem can be found in reference [2].

In addition to the above overview of the program capabilities, this report presents the input requirements, the output capabilities, details

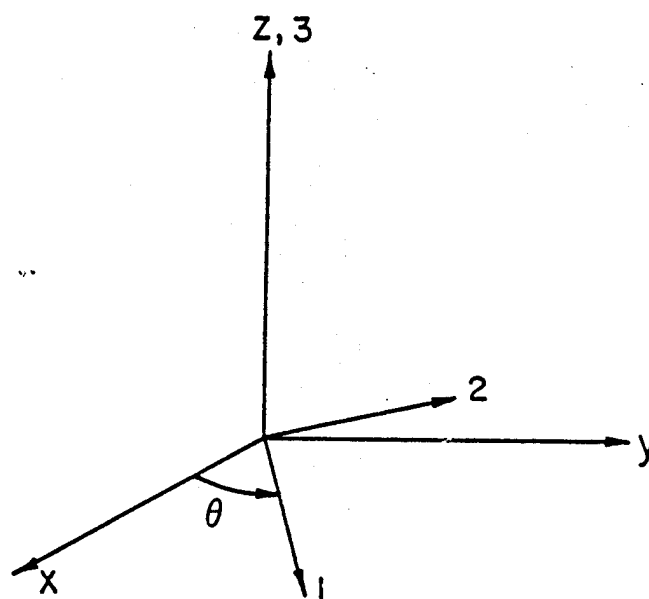


Fig. 1.1 The Coordinate Systems of CLFE2D

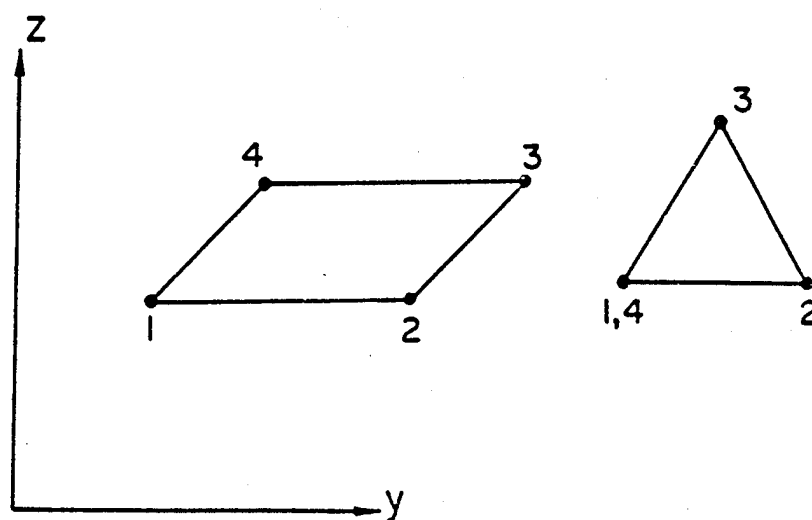


Fig. 1.2 Available Element Geometries

of the input card sequence, a note on symmetry requirements and an example problem.

2.0 CLFE2D Input Requirements:

In this section a description of each type of card image in the input file is given. The required format of each card image is given in section 3.0.

Type 1 Title (Cards 1 & 2)

The title consists of two 80 character lines which will be printed on the output file.

Type 2 Master control Information (Card 3)

On this card the user specifies the number of elements and nodes in the mesh, the number of different angles of material principal orientation, the number of specified nodal degrees of freedom and forces. The user also specifies the desired plot option and a data check option on this card (see section 3.0). The data check option permits the user to check the input without proceeding to the solution.

Type 3 Nodal Coordinate Scale Factors (Card 4)

To ease input the user can specify scale factors which independently scale the magnitude of the Y and Z nodal coordinate values.

Type 4 Printer Control Keys (Card 5)

On this card the user sets the various output keys (see section 3.0) which allow for the generation of a tailored output file.

Type 5 Plotter Control Information (Card 6)

On this card the user specifies the plotter scale factors (see

Output Capabilities). Card 6 can be omitted if output plots are not desired.

Type 6 Material Property Information (Card 7 and 8)

On these two cards the user inputs one set of three dimensional orthotropic material properties. These properties must include thermal or hygroscopic expansion coefficients if the response to hygrothermal loading is desired.

Type 7 Material Property Orientation Information (Card 9)

The user inputs an angle number and magnitude (degrees) for each different material property orientation in the laminate.

Type 8 Element Data Information (Card 10)

The element connectivity data, material property orientation angle number, and the output points (see section 3.0) for each element are specified on this card. Card 10 is repeated NEM times, where NEM is the number of elements in the mesh.

Type 9 Nodal Data Information (Card 11)

The nodal coordinates are specified on this card. The nodal coordinate values must be greater than or equal to zero. Card 11 is repeated NODS times, where NODS is the number of nodes in the mesh.

Type 10 Specified Nodal Displacement Information (Card 12)

On this card the user identifies the node number and displacement of each constrained degree of freedom. Card 12 is repeated for each degree of freedom that has a specified displacement. If there are no constrained degrees of freedom, Card 12 is omitted from the input file.

Type 11 Specified Nodal Force Information (Card 13)

On this card the user identifies the node number and the component magnitude of each of the specified nodal forces. Card 13 is repeated for each component of the specified nodal forces. If there are no specified nodal forces, Card 13 is omitted from the input file.

Type 12 Normal Strain and Hygrothermal Load Information (Card 14)

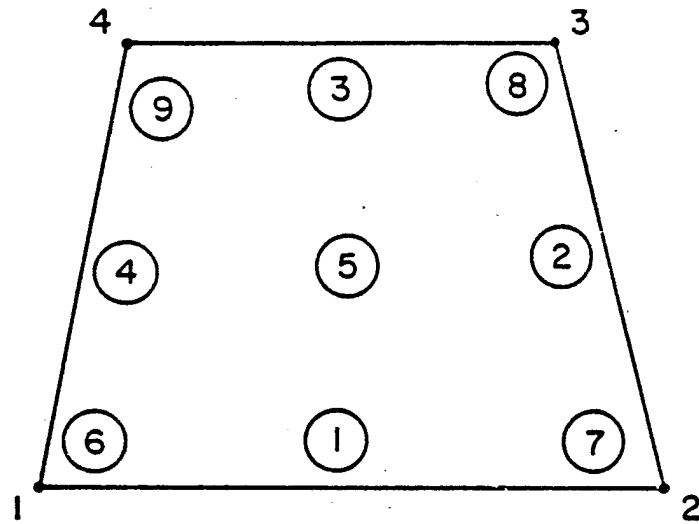
Here the user specifies the applied normal strain ϵ_x^0 along with the temperature or moisture change. It is possible to specify these values as zero.

3.0 CLFE2D Output Capabilities:

CLFE2D solves for the nodal displacements, strains, stresses and strain energy densities for each element in the mesh. These values are calculated in the global (x, y, z) coordinate system. For a given element, the values of strain, stress and strain energy density are calculated at a combination of any nine output points. These output points are specified for each element on Card 10 of the input file. Figure (1.3) illustrates the location of these output points on a given element. Output point 5 is the only valid output point for a triangular element. Specifying other values for triangular elements can cause errors.

It should be noted that the strains, stresses, and strain energies that are printed are the response to the combined mechanical and Hygrothermal loadings. The u displacement that is printed is the $U(y,z)$ of eqn. (1a). To obtain the total u displacement at a desired x location, one must take into account the applied normal strain ϵ_x^0 .





• Element local node number

(n) Location of output point n

Fig. 1.3 Element Stress-Strain Output Points

In CLFE2D the user has the ability to tailor the output file. By setting various printer control keys on Card 5 of the input file, he can opt to have all or any of the following printed:

Key (1)	Element Data
Key (2)	Nodal Data
Key (3)	Specified Nodal Displacements
Key (4)	Specified Nodal Forces
Key (5)	Displacements
Key (6)	Strains, Stresses and Strain Energy Densities

The user can also opt to plot the undeformed and deformed meshes by selecting the appropriate plot option on Card 3. The user controls the scale of these plots by specifying various plot scale factors on Card 6. The various plot options and scale factors are described in detail in the following outline.

1. Plot Option NPLLOT (Card 3)

NPLLOT is one of the variables which is user specified on Card 3.

The user determines which plots will be printed by setting NPLLOT to any of the following values.

0 = No plots

1 = Undeformed plot only

2 = Deformed plot only

3 = Both undeformed and deformed plots

2. Plotter Control Information (Card 6)

On Card 6 of the input file, the user specifies the following plot scale factors:

(a) PYSCL = Plot Y-direction scale factor

This variable determines the maximum length of the plot in the Y-direction. PYSCL must be chosen such that the maximum Y-coordinate in the mesh will be scaled to a value less than or equal to 18 inches.

(b) PZSCL = Plot Z-direction scale factor

This variable determines the maximum length of the plot in the Z-direction. PZSCL must be chosen such that the maximum Z-coordinate in the mesh will be scaled to a value less than or equal to 9 inches.

(c) VMAX = Maximum allowable v-displacement

This variable determines the size of the maximum v-displacement in inches. All of the smaller v-displacements are scaled proportionally to this value.

(d) WMAX = Maximum allowable w-displacement

This variable determines the size of the maximum w-displacement in inches. All of the smaller w-displacements are scaled proportionally to this value.

4.0 CLFE2D Input Card Sequence

<u>Cards 1 & 2 (20A4)</u>		<u>Title Cards</u>
<u>Column</u>		<u>Contents</u>
1-80		Title
<u>Card 3 (1615)</u>		<u>Master Control Card</u>
<u>Column</u>		<u>Contents</u>
1-5	NEM	Number of Elements
6-10	NODS	Number of Node Points
11-15	NANG	Number of Different Angles
16-20	NSDF	Number of Specified Nodal Displacements
21-25	NSBF	Number of Specified Nodal Forces
26-30	NPLOT	Plot Option
		= 0 none
		= 1 undeformed plot only
		= 2 deformed plot only
		= 3 both undeformed and deformed
41-45	NCHECK	Data Check Option

By setting NPLOT .EQ.1 and NCHECK .NE.0 a data check and an undeformed plot will be generated. To obtain the complete solution, another run must be made with NCHECK .EQ.0.

<u>Card 4 (8D10.5)</u>		<u>Scale Factors</u>
<u>Column</u>		<u>Contents</u>
1-10	SCAY	Y-Nodal Coordinate Scale Factor
11-20	SCAZ	Z-Nodal Coordinate Scale Factor
<u>Card 5 (1615)</u>		<u>Printer Control Card</u>
<u>Column</u>		<u>Contents</u>
1-5	KEY(1)	Key for Printing Element Data
6-10	KEY(2)	Key for Printing Nodal Data
11-15	KEY(3)	Key for Printing Specified Nodal Displacements
16-20	KEY(4)	Key for Printing Specified Nodal Forces
21-25	KEY(5)	Key for Printing Displacements
26-30	KEY(6)	Key for Printing Strains; Stresses and Strain Energies Per Unit Volume
		(if KEY(i).NE. 0 - Print)

(NPLOT .EQ. 0 skip Card 6 and go to Card 7)

<u>Card 6 (8E10.5)</u>		<u>Plotter Control Card</u>
<u>Column</u>		<u>Contents</u>
1-10	PYSCL	Plot Y-Scale Factor
11-20	PZSCL	Plot Z-Scale Factor
21-30	VMAX	Maximum Allowable V-Displacement
31-40	WMAX	Maximum Allowable W-Displacement

Card 7 (8D10.5)

Material Properties Card A

<u>Column</u>		<u>Contents</u>
1-10	PROP (1)	E-11
11-20	PROP (2)	E-22
21-30	PROP (3)	E-33
31-40	PROP (4)	G-23
41-50	PROP (5)	G-13
51-60	PROP (6)	G-12

Card 8 (8D10.5)

Material Properties Card 8

Column	Contents
1-10	Prop(7) NU-23
11-20	Prop(8) NU-13
21-30	Prop(9) NU-12
31-40	Prop(10) Alpha-11
41-50	Prop(11) Alpha-22
51-60	Prop(12) Alpha-33

Card 9 (8D10.5)

Angle Data Card

Column	Contents	
1-10	Ang(1)	Angle No. 1 (in Degrees)
11-20	Ang(2)	Angle No. 2 (in Degrees)
.	.	.
.	.	.
.	.	.
.	Ang(NANG)	Angle No. NANG (in Degrees)

Card 10 (5X,6I5)

Element Data Card(s)

Column	Contents
1-5	Blank (Element Numbers may be inserted for Reference)
6-10	NOD(N,1) Node #1 of Element N
11-15	NOD(N,2) Node #2 of Element N
16-20	NOD(N,3) Node #3 of Element N
21-25	NOD(N,4) Node #4 of Element N
26-30	IANG(N) Angle Number of Element N
31-35	ISTRS(N,1) Stress and Strain Output Code for Element N. = 0 - None = 1 - Side 1 = 2 - Side 2 = 3 - Side 3 = 4 - Side 4 = 5 - Center = 6 - "Node 1" = 7 - "Node 2" = 8 - "Node 3" = 9 - "Node 4"
36-40	ISTRS(N,2)
41-45	ISTRS(N,3)

36-40	ISTRS(N,2)
41-45	ISTRS(N,3)
.	.
.	.
.	.

71-75 ISTRS(N,9)
 *Repeat Card 10 NEM Times

<u>Card 11 (5X,2D10.5)</u>		Nodal Data Card(s)
<u>Column</u>		<u>Contents</u>
1-5	Blank	(Node Numbers may be inserted for Reference)
6-15	Y(N)	Y-Coordinate of Node N (This value must be > zero)
16-25	Z(N)	Z-Coordinate of Node N (This value must be > zero)

*Repeat Card 11 NODS Times #

#If NSDF .EQ. 0 - Skip Card 12 and Go on to Card 13

<u>Card 12 (2I5,D10.5)</u>		Specified Nodal Degree of Freedom Card(s)
<u>Column</u>		<u>Contents</u>
1-5	ND	Node Number of Specified D.O.F.
6-10	IDR	Direction of Specified D.O.F. (1 = u, 2 = v, and 3 = w)
11-20	VBDF(N)	Specified Displacement Value

*Repeat Card 12 NSDF Times

* If NSBF .EQ. 0 - Skip Card 13 and Go on to Card 14

<u>Card 13 (2I5,D10.5)</u>		Specified Nodal Force Card(s)
<u>Column</u>		<u>Contents</u>
1-5	ND	Node Number of Specified Force
6-10	IDR	Direction of Specified Force (1 = u, 2 = v, 3 = w)
11-20	VBSF(N)	Specified Boundary Force Value

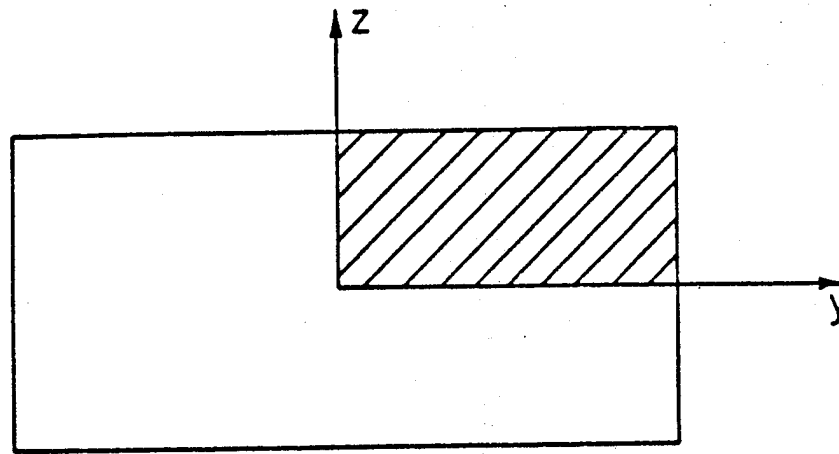
*Repeat Card 13 NSBF Times

<u>Card 14 (8D10.5)</u>		Normal Strain and Hygrothermal Load Card
<u>Column</u>		<u>Contents</u>
1-10	EPSX	Applied Normal, (X-Direction), Strain
11-20	TEMP	Temperature Change or Moisture Concentration

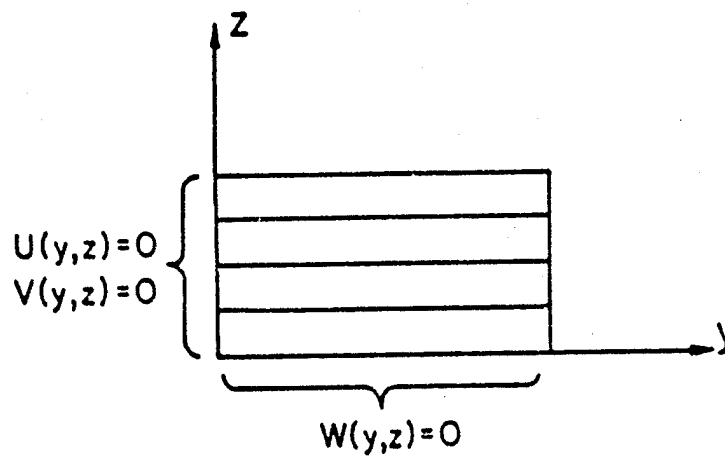
-End-

5.0 A Note on Symmetry Requirements

To reduce the cost of a finite element stress analysis symmetry arguments are often used when building the finite element mesh. To determine the response of a thin composite laminate tensile specimen the analyst need only model one quarter of the specimen cross-section. To obtain valid results, however, the appropriate symmetry boundary conditions must be specified. These boundary conditions are shown in Fig. (1.4). Along the z -axis ($y=0$) the U and V nodal displacements must be specified to be zero. Similarly, along the y -axis ($z=0$) the W displacements must be set equal to zero.



The upper right quadrant of the specimen cross-section will be modeled using CLFE2D



The appropriate boundary conditions for the quarter symmetry model are shown above

Fig. 1.4 Appropriate Boundary Conditions for a Quarter Symmetry Model

6.0 Example Problem

Consider an eight ply T300/5208 Graphite Epoxy laminate with a $[+45_2/-45_2]_s$ stacking sequence. The laminate is 0.50 inches wide and each ply is 0.005 inches thick. It is desired to study the response of the laminate to an applied tensile normal strain $\epsilon_x^0 = 0.001$. Of particular interest are the stress, strain and strain energy density distributions along the $+45^\circ/-45^\circ$ interface.

Solution: A 36 element, 42 node quarter symmetry mesh was developed to study the given problem using CLFE2D. The input file and output for this problem are shown in Appendix A. Note that the stress strain output code varies for each element in the mesh. By doing this, the overall response of the laminate and the response at the $+45^\circ, -45^\circ$ interface can be obtained with a minimum of output.

A simple finite element mesh was used in the solution of the example problem for the purpose of exposition. However, CLFE2D is capable of solving large scale research problems.

REFERENCES

1. Segerlind, L. J., Applied Finite Element Analysis, John Wiley & Sons, Inc., New York, 1976, pp. 80-84, 314-316.
2. Renieri, G. D., Herakovich, C. T., "Nonlinear Analysis of Laminated Fibrous Composites," VPI-E-76-10, College of Engineering, Virginia Polytechnic Institute & State University, Blacksburg, VA, 1976.

APPENDIX

EXAMPLE INPUT DATA

EXAMPLE PROBLEM, T300/5208, 8-LAYER, (45.0, 45.0, -45.0, -45.0)S

EPSX= 0.001, HYGTROTHERMAL LOAD = 0.000

36	42	4	16	0	3	0	0	0	0
0.01	2.50D-03								
1	1	1	1	1	1				
32.0	100.0			0.5	0.5				
.1920D+08	1.560D+06	1.560D+06	.4870D+06	.8200D+06	.8200D+06				
0.490	0.238	0.238	0.00	.1380D-04	.1380D-04				
45.0	45.0	-45.0	-45.0						
1	4	9	10	5	1	4			
2	3	8	9	4	2	4	6	1	7
3	2	7	8	3	3	4	9	3	8
4	1	6	7	2	4	4			
5	9	14	15	10	1				
6	8	13	14	9	2	6	1	7	
7	7	12	13	8	3	9	3	8	
8	6	11	12	7	4				
9	15	23	24	15	1	5			
10	14	23	15	14	1	5			
11	14	22	23	14	1	5			
12	14	21	22	14	2	5			
13	13	21	14	13	2	5			
14	13	20	21	13	2	5			
15	13	19	20	13	3	5			
16	12	19	13	12	3	5			
17	12	18	19	12	3	5			
18	12	17	18	12	4	5			
19	11	17	12	11	4	5			
20	11	16	17	11	4	5			
21	23	32	33	24	1				
22	22	31	32	23	1				
23	21	30	31	22	2				
24	20	29	30	21	2	6	1	7	
25	19	28	29	20	3	9	3	8	
26	18	27	28	19	3				
27	17	26	27	18	4				
28	16	25	26	17	4				
29	32	41	42	33	1	5	2		
30	31	40	41	32	1	5			
31	30	39	40	31	2	5			
32	29	38	39	30	2	5	6	1	7
33	28	37	38	29	3	5	9	3	8
34	27	36	37	28	3	5			
35	26	35	36	27	4	5			
36	25	34	35	26	4	5			
1		0.0		0.0					
2		0.0		2.0					
3		0.0		4.0					
4		0.0		6.0					
5		0.0		8.0					
6		8.0		0.0					

WATER QUALITY
 PERSON QUALITY

7	8.0	2.0
8	8.0	4.0
9	8.0	6.0
10	8.0	8.0
11	16.0	0.0
12	16.0	2.0
13	16.0	4.0
14	16.0	6.0
15	16.0	8.0
16	20.0	0.0
17	20.0	1.0
18	20.0	2.0
19	20.0	3.0
20	20.0	4.0
21	20.0	5.0
22	20.0	6.0
23	20.0	7.0
24	20.0	8.0
25	23.0	0.0
26	23.0	1.0
27	23.0	2.0
28	23.0	3.0
29	23.0	4.0
30	23.0	5.0
31	23.0	6.0
32	23.0	7.0
33	23.0	8.0
34	25.0	0.0
35	25.0	1.0
36	25.0	2.0
37	25.0	3.0
38	25.0	4.0
39	25.0	5.0
40	25.0	6.0
41	25.0	7.0
42	25.0	8.0

1	1	0.0
2	1	0.0
3	1	0.0
4	1	0.0
5	1	0.0
1	2	0.0
2	2	0.0
3	2	0.0
4	2	0.0
5	2	0.0
1	3	0.0
6	3	0.0
11	3	0.0
16	3	0.0
25	3	0.0
34	3	0.0

0.001

0.00

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EXAMPLE OUTPUT

*** TWO DIMENSIONAL ***
*** COMPOSITE LAMINATE FINITE ELEMENTS ***

LAYER THICKNESS: 1.000E+00, 8-LAYER, (0.0, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0)
TENS. 0.001, INITIAL TENS. 0.000

***** COMPOSITE INFORMATION *****

NUMBER OF ELEMENTS	36
NUMBER OF NODES	96
NUMBER OF DIFFERENT ANGLES	4
NUMBER OF SELECTED DEGREES OF FREEDOM	16
NUMBER OF SELECTED BOUNDARY FORCES	0
NUMBER OF SELECTED DISPLACEMENTS	3
NUMBER OF SELECTED STRESSES	0
NUMBER OF SELECTED STRAINS	0
NUMBER OF SELECTED THERMAL STRESSES	0
DATA ERROR DETECTION	0

***** SCALE FACTORS *****

1 - SCALE FACTOR	0.100000E-01
2 - SCALE FACTOR	0.250000E-02

***** PRINTING CONTROL OPTIONS *****

1 - FOR PRINTING LAMINATE DATA	1
2 - FOR PRINTING NODE DATA	1
3 - FOR PRINTING BOUNDARY DISPLACEMENTS	1
4 - FOR PRINTING BOUNDARY FORCES	1
5 - FOR PRINTING NODE DISPLACEMENTS	1
6 - FOR PRINTING STRESSES AND STRAINS	1

***** PRINTING CONTROL INFORMATION *****

1 - FOR PRINTING LAMINATE DATA	0.100000E-01
2 - FOR PRINTING NODE DATA	0.250000E-02
3 - FOR PRINTING BOUNDARY DISPLACEMENTS	0.500000E-03
4 - FOR PRINTING BOUNDARY FORCES	0.500000E-03
5 - FOR PRINTING NODE DISPLACEMENTS	0.500000E-03
6 - FOR PRINTING STRESSES AND STRAINS	0.500000E-03

***** MATERIAL PROPERTIES *****

COMPOSITE INFORMATION *****

1 - FOR PRINTING LAMINATE DATA	0.100000E-01
2 - FOR PRINTING NODE DATA	0.250000E-02
3 - FOR PRINTING BOUNDARY DISPLACEMENTS	0.500000E-03
4 - FOR PRINTING BOUNDARY FORCES	0.500000E-03
5 - FOR PRINTING NODE DISPLACEMENTS	0.500000E-03
6 - FOR PRINTING STRESSES AND STRAINS	0.500000E-03

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ARE POOR ORIGINALS

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0.1 0.457000000000
0.2 0.820000000000
0.3 0.820000000000

FUSION RATIOS

0.1 0.420000000000
0.2 0.230000000000
0.3 0.230000000000

Coefficients of Thermal Expansion

ALPH1 0.0
ALPH2 0.13000000-06
ALPH3 0.11800000-06

• ANGLE INFORMATION •

ANGLE NO. DATA IN DEGREES

1 0.450000000000
2 0.450000000000
3 -0.450000000000
4 -0.450000000000

• THERMAL DATA •

TEMP	MOD-1	MOD-2	MOD-3	MOD-4	ANGLE	STRESS OPTIONS
1	4	9	10	5	1	0 0 0 0 0 0 0 0
2	3	8	9	4	2	0 6 1 7 0 0 0 0
3	2	7	8	3	3	0 9 3 8 0 0 0 0
4	1	6	7	2	4	0 0 0 0 0 0 0 0
5	9	15	16	10	1	0 0 0 0 0 0 0 0
6	8	14	15	9	2	0 1 7 0 0 0 0 0
7	7	13	14	8	3	0 1 8 0 0 0 0 0
8	6	12	13	7	4	0 0 0 0 0 0 0 0
9	5	11	12	6	1	0 0 0 0 0 0 0 0
10	15	24	25	15	1	0 0 0 0 0 0 0 0
11	14	23	24	14	1	0 0 0 0 0 0 0 0
12	13	22	23	13	1	0 0 0 0 0 0 0 0
13	12	21	22	12	2	0 0 0 0 0 0 0 0
14	11	20	21	11	2	0 0 0 0 0 0 0 0
15	10	19	20	10	3	0 0 0 0 0 0 0 0
16	9	18	19	9	3	0 0 0 0 0 0 0 0
17	8	17	18	8	4	0 0 0 0 0 0 0 0
18	7	16	17	7	4	0 0 0 0 0 0 0 0
19	6	15	16	6	1	0 0 0 0 0 0 0 0
20	5	14	15	5	1	0 0 0 0 0 0 0 0
21	4	13	14	4	1	0 0 0 0 0 0 0 0
22	3	12	13	3	2	0 0 0 0 0 0 0 0
23	2	11	12	2	2	0 0 0 0 0 0 0 0
24	1	10	11	1	3	0 0 0 0 0 0 0 0

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• PHALIA •

Run No.	Z-COORDINATE	Z-COORDINATE
1	0.0	0.0
2	0.0	0.5000000D-02
3	0.0	0.1000000D-01
4	0.0	0.1500000D-01
5	0.0	0.2000000D-01
6	0.8000000D-01	0.0
7	0.8000000D-01	0.5000000D-02
8	0.8000000D-01	0.1000000D-01
9	0.8000000D-01	0.1500000D-01
10	0.8000000D-01	0.2000000D-01
11	0.1600000D-02	0.0
12	0.1600000D-02	0.5000000D-02
13	0.1600000D-02	0.1000000D-01
14	0.1600000D-02	0.1500000D-01
15	0.1600000D-02	0.2000000D-01
16	0.2000000D-02	0.0
17	0.2000000D-02	0.5000000D-02
18	0.2000000D-02	0.1000000D-01
19	0.2000000D-02	0.1500000D-01
20	0.2000000D-02	0.2000000D-01
21	0.2000000D-01	0.0
22	0.2000000D-01	0.5000000D-02
23	0.2000000D-01	0.1000000D-01
24	0.2000000D-01	0.1500000D-01
25	0.2000000D-01	0.2000000D-01
26	0.2500000D-02	0.0
27	0.2500000D-02	0.5000000D-02
28	0.2500000D-02	0.1000000D-01
29	0.2500000D-02	0.1500000D-01
30	0.2500000D-01	0.0
31	0.2500000D-01	0.5000000D-02
32	0.2500000D-01	0.1000000D-01
33	0.2500000D-01	0.1500000D-01
34	0.2500000D-01	0.2000000D-01
35	0.2500000D-02	0.0
36	0.2500000D-02	0.5000000D-02
37	0.2500000D-02	0.1000000D-01
38	0.2500000D-02	0.1500000D-01
39	0.2500000D-01	0.0

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• UNIFORMED FLOI COMPLET •

60	0.250000000000000000	0.150000000000000000
61	0.250000000000000000	0.175000000000000000
62	0.250000000000000000	0.200000000000000000

• SPECIFIED DIRECTIONS OF FREEDOM •

MODE NO.	DIRECTION	SPECIFIED DISPLACEMENT
1	1	0.0
2	1	0.0
3	1	0.0
4	1	0.0
5	1	0.0
6	1	0.0
7	2	0.0
8	2	0.0
9	2	0.0
10	2	0.0
11	3	0.0
12	3	0.0
13	3	0.0
14	3	0.0
15	3	0.0

• STRAIN AND INTERNAL LOADS •

TEMPERATURE CHANGE	0.100000000000000000
TEMPERATURE CHANGE	0.0

• DETACHED FLOI COMPLET •

DISPLACEMENT SCALE FACTORS AND V	0.27010531400 W	0.1105231406
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• MODAL DISPLACEMENTS •

MODE NO.	U-DISPLACEMENT	V-DISPLACEMENT	W-DISPLACEMENT
1	0.0	0.0	0.0
2	0.0	0.0	-0.56206910-06
3	0.0	0.0	-0.11293070-05
4	0.0	0.0	-0.16867070-05
5	0.0	0.0	-0.22490790-05
6	0.57067360-09	-0.58907190-04	0.0
7	0.62220650-09	-0.58907630-04	-0.56307560-06
8	0.53613950-09	-0.58908070-04	-0.11256000-05
9	0.45659230-09	-0.58908900-04	-0.16878190-05
10	0.39533180-09	-0.58909250-04	-0.22500630-05
11	-0.11507970-08	-0.11507970-03	0.0

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12	-0.1553664D+08	-0.1178192D+03	-0.5589134D+06
13	-0.1675590D+08	-0.1178183D+03	-0.119910D+05
14	-0.1695179D+08	-0.1178157D+03	-0.1683737D+05
15	-0.1751953D+08	-0.1178150D+03	-0.22945067D+05
16	-0.4628679D+07	-0.1473319D+03	0.0
17	-0.4212072D+07	-0.1473334D+03	-0.2686268D+06
18	-0.3376679D+07	-0.1473360D+03	-0.5008633D+06
19	-0.2512133D+07	-0.1473394D+03	-0.8187789D+06
20	-0.2331912D+07	-0.1472748D+03	-0.1102637D+05
21	-0.2395740D+07	-0.1472762D+03	-0.1393398D+05
22	-0.4466680D+07	-0.1472753D+03	-0.1674805D+05
23	-0.3309286D+08	-0.1472716D+03	-0.1951650D+05
24	-0.8505363D+09	-0.1472747D+03	-0.2227739D+05
25	0.0300415D+06	-0.1691959D+03	0.0
26	0.7874630D+06	-0.1690996D+03	-0.4027783D+06
27	0.5976040D+06	-0.1690966D+03	-0.7889364D+06
28	0.1430451D+06	-0.1689963D+03	-0.1743992D+05
29	-0.946866D+05	-0.1689938D+03	-0.1453855D+05
30	-0.3308369D+06	-0.1688518D+03	-0.3174166D+05
31	-0.4394723D+06	-0.1638395D+03	-0.2178083D+05
32	-0.9680237D+06	-0.1687880D+03	-0.23127411D+05
33	-0.1065715D+05	-0.1638213D+03	-0.2913377D+05
34	0.7065500D+05	-0.1538213D+03	0.0
35	0.670004D+05	-0.1539856D+03	-0.59614451D+06
36	0.5565810D+05	-0.181376D+03	-0.1177389D+05
37	0.3460561D+05	-0.1813306D+03	-0.1701912D+05
38	-0.1300830D+06	-0.1874116D+03	-0.2294746D+05
39	-0.3700470E+05	-0.1811806D+03	-0.2598058D+05
40	-0.5876272D+05	-0.1801958D+03	-0.3142730D+05
41	-0.6978670D+05	-0.1795353D+03	-0.3742847D+05
42	-0.7340360D+05	-0.1791830D+03	-0.43727451D+05

CURRENT STRESSES AND STRAINS IN GLOBAL X - Y - Z COORDINATES *

ELEMENT ()		STRESSS, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (h)		COORDINATES OF OUTPUT POINTS	
1PSX8	0.100000D+02	S1GX8	-0.284760D+04	Y	0.0
1PSY4	-0.136363D+03	S1GY4	-0.334578D+01	Z	0.150000D+01
1PSZ7	-0.1124674D+03	S1GZ7	-0.108936D+01		
6GMY4	-0.1148593D+01	LAUY4	-0.8520953D+02		
6GMY7	0.0	LAUZ7	-0.104507D+02		
6GMY8	0.54076569D+08	LAUX8	0.116804D+04	STRAIN ENERGY	0.102381D+01

TERMID	(σ) STRESSES	STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (b)	COORDINATES OF OUTPUT POINTS
			X Y Z
EPSX	0.100000D-02	SIGXX	0.2807633D+04
EPSY	-0.1363561D-03	SIGYY	0.9370057D-02
EPSZ	-0.115700D-03	SIGZZ	-0.3763867D-01
GAMX	-0.1456678D-07	TAUZY	-0.9519391D-02
GAMY	0.0	TAUZX	-0.2929536D-02
GAMZ	0.62006618D-08	TAUZY	0.1168070D+04
		STRAIN ENERGY	0.1022849D+01

TEMPERATURE (°C)	STRESS, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (6)	COORDINATES OF ORIGIN POINTS
	STRESS	x, y, z
TEMPERATURE (°C)	0.1000000-02	0.200/6610+04
STRESS	-0.7363000-03	0.200/6610+04
STRAIN	-0.1190900-03	0.200/6610+04
STRAIN ENERGY	-0.1190900-03	0.200/6610+04

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* ELEMENT (2) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (1) *			* ELEMENT (2) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (1) *		
CAHZ	-0.1566960D-07	IAUZY	-0.1024000D-01	COORDINATES OF OUTPUT POINTS	
CABZ	0.0	IAUXZ	-0.2608980D-02	Y	0.4000000D-01
CABY	0.6701740D-08	IAUXY	0.1168100D+00	Z	0.1000000D-01
STRAIN ENERGY = 0.1423819D+01			STRAIN ENERGY = 0.1423819D+01		
* ELEMENT (3) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (1) *			* ELEMENT (3) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (1) *		
IPSSX	0.1000000D-02	SIGXX	0.2847670D+00	COORDINATES OF OUTPUT POINTS	
IPSSY	-0.7363509D-03	SIGYY	0.6103990D-01	Y	0.4000000D-01
IPSSZ	-0.1124614D-03	SIGZZ	0.8565720D-02	Z	0.1000000D-01
CABZ	-0.9781530D-07	IAUZY	-0.6246600D-01		
CABX	-0.7950010D-08	IAUXZ	-0.2708420D-01		
CABY	0.6701740D-08	IAUXY	0.1168100D+00	STRAIN ENERGY = 0.1423819D+01	
STRAIN ENERGY = 0.1423819D+01			STRAIN ENERGY = 0.1423819D+01		
* ELEMENT (3) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (1) *			* ELEMENT (3) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (1) *		
IPSSX	0.1000000D-02	SIGXX	0.2847670D+00	COORDINATES OF OUTPUT POINTS	
IPSSY	-0.7363509D-03	SIGYY	0.6103990D-01	Y	0.4000000D-01
IPSSZ	-0.1124614D-03	SIGZZ	0.8565720D-02	Z	0.1000000D-01
CABZ	-0.9781530D-07	IAUZY	-0.6246600D-01		
CABX	-0.7950010D-08	IAUXZ	-0.2708420D-01		
CABY	0.6701740D-08	IAUXY	0.1168100D+00	STRAIN ENERGY = 0.1423819D+01	
STRAIN ENERGY = 0.1423819D+01			STRAIN ENERGY = 0.1423819D+01		
* ELEMENT (3) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (1) *			* ELEMENT (3) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (1) *		
IPSSX	0.1000000D-02	SIGXX	0.2847670D+00	COORDINATES OF OUTPUT POINTS	
IPSSY	-0.7363509D-03	SIGYY	0.6103990D-01	Y	0.4000000D-01
IPSSZ	-0.1124614D-03	SIGZZ	0.8565720D-02	Z	0.1000000D-01
CABZ	-0.9781530D-07	IAUZY	-0.6246600D-01		
CABX	-0.7950010D-08	IAUXZ	-0.2708420D-01		
CABY	0.6701740D-08	IAUXY	0.1168100D+00	STRAIN ENERGY = 0.1423819D+01	
STRAIN ENERGY = 0.1423819D+01			STRAIN ENERGY = 0.1423819D+01		
* ELEMENT (3) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (1) *			* ELEMENT (3) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (1) *		
IPSSX	0.1000000D-02	SIGXX	0.2847670D+00	COORDINATES OF OUTPUT POINTS	
IPSSY	-0.7363509D-03	SIGYY	0.6103990D-01	Y	0.4000000D-01
IPSSZ	-0.1124614D-03	SIGZZ	0.8565720D-02	Z	0.1000000D-01
CABZ	-0.9781530D-07	IAUZY	-0.6246600D-01		
CABX	-0.7950010D-08	IAUXZ	-0.2708420D-01		
CABY	0.6701740D-08	IAUXY	0.1168100D+00	STRAIN ENERGY = 0.1423819D+01	
STRAIN ENERGY = 0.1423819D+01			STRAIN ENERGY = 0.1423819D+01		
* ELEMENT (3) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (1) *			* ELEMENT (3) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (1) *		
IPSSX	0.1000000D-02	SIGXX	0.2847670D+00	COORDINATES OF OUTPUT POINTS	
IPSSY	-0.7363509D-03	SIGYY	0.6103990D-01	Y	0.4000000D-01
IPSSZ	-0.1124614D-03	SIGZZ	0.8565720D-02	Z	0.1000000D-01
CABZ	-0.9781530D-07	IAUZY	-0.6246600D-01		
CABX	-0.7950010D-08	IAUXZ	-0.2708420D-01		
CABY	0.6701740D-08	IAUXY	0.1168100D+00	STRAIN ENERGY = 0.1423819D+01	
STRAIN ENERGY = 0.1423819D+01			STRAIN ENERGY = 0.1423819D+01		

W. A. A. . . .

ITEM (9) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (5) •

ITEM	0.100000D-02	SIGXX	0.280750D+00	COORDINATES OF OUTPUT POINTS
ITEM	-0.106178D+03	SIGYY	0.201675D+00	Y 0.160000D+00
ITEM	-0.112200D+03	SIGZZ	0.525015D+00	Z 0.100000D+01
CARX	0.226615D+06	IAUX	0.158615D+00	
CARY	-0.297596D+07	IAUX	-0.550469D+01	
CARZ	-0.276638D+07	IAUX	-0.116809D+00	STRAIN ENERGY 0.102389D+01

ITEM (9) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (5) •

ITEM	0.100000D-02	SIGXX	0.285227D+00	COORDINATES OF OUTPUT POINTS
ITEM	-0.109800D+03	SIGYY	0.580673D+01	Y 0.180000D+00
ITEM	-0.110791D+03	SIGZZ	0.510155D+01	Z 0.191750D+01
CARX	0.219716D+05	IAUX	0.115268D+01	
CARY	0.167992D+05	IAUX	0.146369D+01	
CARZ	0.678157D+07	IAUX	0.117016D+00	STRAIN ENERGY 0.102387D+01

ITEM (10) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (5) •

ITEM	0.100000D-02	SIGXX	0.280750D+00	COORDINATES OF OUTPUT POINTS
ITEM	-0.105900D+03	SIGYY	0.267292D+01	Y 0.170000D+00
ITEM	-0.112590D+03	SIGZZ	0.2500613D+00	Z 0.168750D+01
CARX	0.496907D+06	IAUX	0.110127D+00	
CARY	0.866357D+07	IAUX	0.135266D+00	
CARZ	-0.516678D+07	IAUX	0.116976D+00	STRAIN ENERGY 0.102375D+01

ITEM (11) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (5) •

ITEM	0.100000D-02	SIGXX	0.2850013D+00	COORDINATES OF OUTPUT POINTS
ITEM	-0.108885D+03	SIGYY	0.317516D+01	Y 0.180000D+00
ITEM	-0.110747D+03	SIGZZ	0.405666D+01	Z 0.156250D+01
CARX	0.652714D+07	IAUX	0.199982D+00	
CARY	0.462956D+05	IAUX	0.296766D+01	
CARZ	-0.329116D+06	IAUX	0.316817D+00	STRAIN ENERGY 0.102393D+01

ITEM (12) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (5) •

ITEM	0.100000D-02	SIGXX	0.2808913D+00	COORDINATES OF OUTPUT POINTS
ITEM	-0.108885D+03	SIGYY	0.1955915D+01	Y 0.180000D+00
ITEM	-0.112622D+03	SIGZZ	0.257131D+00	Z 0.103750D+01
CARX	0.275181D+05	IAUX	0.245269D+01	
CARY	0.5718638D+05	IAUX	0.409577D+01	
CARZ	-0.329116D+06	IAUX	0.116809D+00	STRAIN ENERGY 0.102399D+01

ITEM (13) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (5) •

ITEM	0.100000D-02	SIGXX	0.280697D+00	COORDINATES OF OUTPUT POINTS
ITEM	-0.105978D+03	SIGYY	-0.527131D+00	Y 0.170000D+00
ITEM	-0.112991D+03	SIGZZ	0.371197D+00	Z 0.118750D+01
CARX	0.157566D+06	IAUX	0.473703D+00	
CARY	0.160160D+07	IAUX	0.142709D+00	
CARZ	-0.659306D+06	IAUX	0.116675D+00	STRAIN ENERGY 0.102391D+01

ITEM (14) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (5) •

ITEM	0.100000D-02	SIGXX	0.286259D+00	COORDINATES OF OUTPUT POINTS
ITEM	-0.106930D+03	SIGYY	-0.507577D+00	Y 0.180000D+00

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EPSX	-0.115106D-01	SIGZZ	-0.5464958D+01	Z	0.1062500D-01
EPZY	-0.1789998D-05	TAUZZ	0.4114899D+01	STRAIN ENERGY	0.1625106D+01
EPXZ	-0.2255113D-05	TAUZY	-0.1769062D+00	VOLUME EVALUATED AT POINT (%)	
EPYZ	-0.5610265D-06	TAUZY	0.1165668D+00	COORDINATES OF OUTPUT POINTS	
				Y	0.1800000D+00
				Z	0.9375000D-02
* ELEMENT (15) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (%) *					
EPSX	0.100000D-02	SIGXX	0.2847672D+00	STRAIN ENERGY	0.1629057D+01
EPZY	-0.1365106D-03	SIGYY	-0.4182933D-01	VOLUME EVALUATED AT POINT (%)	
EPXZ	-0.11917611D-03	SIGZZ	-0.4912979D+01	COORDINATES OF OUTPUT POINTS	
EPYZ	0.9700013D-05	TAUXX	0.2958329D+01	Y	0.1800000D+00
EPXZ	-0.1235330D-05	TAUYY	-0.1511308D+01	Z	0.6875000D-02
EPYZ	-0.5610265D-06	TAUZZ	-0.1170870D+00	STRAIN ENERGY	
* ELEMENT (16) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (%) *					
EPSX	0.100000D-02	SIGXX	0.2808066D+00	STRAIN ENERGY	0.1629762D+01
EPZY	-0.1365106D-03	SIGYY	0.1715725D+00	VOLUME EVALUATED AT POINT (%)	
EPXZ	-0.1125000D-03	SIGZZ	0.2287082D+00	COORDINATES OF OUTPUT POINTS	
EPYZ	0.6829657D-06	TAUXX	0.4501138D+00	Y	0.1800000D+00
EPXZ	-0.2672966D-07	TAUYY	-0.1298110D+00	Z	0.5625000D-02
EPYZ	-0.4679098D-06	TAUZZ	-0.1167092D+00	STRAIN ENERGY	
* ELEMENT (17) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (%) *					
EPSX	0.100000D-02	SIGXX	0.2808066D+00	STRAIN ENERGY	0.1629762D+01
EPZY	-0.1365106D-03	SIGYY	-0.4630915D+00	VOLUME EVALUATED AT POINT (%)	
EPXZ	-0.1111967D-03	SIGZZ	0.1869167D+01	COORDINATES OF OUTPUT POINTS	
EPYZ	0.9759630D-05	TAUXX	0.2966199D+01	Y	0.1800000D+00
EPXZ	-0.2672966D-07	TAUYY	0.1946657D+01	Z	0.5625000D-02
EPYZ	-0.8051833D-06	TAUZZ	-0.1168173D+00	STRAIN ENERGY	
* ELEMENT (18) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (%) *					
EPSX	0.100000D-02	SIGXX	0.2820063D+00	STRAIN ENERGY	0.1629762D+01
EPZY	-0.1371117D-03	SIGYY	0.1035728D+01	VOLUME EVALUATED AT POINT (%)	
EPXZ	-0.1088996D-03	SIGZZ	0.6550114D+01	COORDINATES OF OUTPUT POINTS	
EPYZ	0.2991967D-05	TAUXX	0.5871749D+01	Y	0.1800000D+00
EPXZ	0.1041577D-05	TAUYY	0.1898701D+00	Z	0.1875000D-02
EPYZ	-0.8051833D-06	TAUZZ	-0.1167869D+00	STRAIN ENERGY	
* ELEMENT (19) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (%) *					
EPSX	0.100000D-02	SIGXX	0.2807636D+00	STRAIN ENERGY	0.1629762D+01
EPZY	-0.1371117D-03	SIGYY	-0.4352805D+01	VOLUME EVALUATED AT POINT (%)	
EPXZ	-0.1117470D-03	SIGZZ	-0.1534368D-01	COORDINATES OF OUTPUT POINTS	
EPYZ	0.1876162D-06	TAUXX	0.2633018D+01	Y	0.1800000D+00
EPXZ	-0.8011710D-07	TAUYY	-0.1160723D+00	Z	0.625000D-03
EPYZ	-0.1046590D-05	TAUZZ	-0.1160699D+00	STRAIN ENERGY	
* ELEMENT (20) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (%) *					
EPSX	0.100000D-02	SIGXX	0.2808587D+00	STRAIN ENERGY	0.1629762D+01
EPZY	-0.1390517D-03	SIGYY	-0.1822230D+01	VOLUME EVALUATED AT POINT (%)	
EPXZ	-0.1079507D-03	SIGZZ	0.8760670D+01	COORDINATES OF OUTPUT POINTS	
EPYZ	0.1978960D-05	TAUXX	0.2512816D+01	Y	0.1800000D+00
EPXZ	-0.1760276D-05	TAUYY	0.2512816D+01	Z	0.625000D-03
EPYZ	-0.1760276D-05	TAUZZ	0.2512816D+01	STRAIN ENERGY	

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DATA	-0.117838/D-05	DATA	-0.116583D+00	STRAIN ENERGY	0.162515D+01
• ELEMENT (29) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (6) •					
EPSX	0.100000D-02	SIGXX	0.2910671D+00	COORDINATES OF OUTPUT POINTS Y = 0.200000D+00 Z = 0.100000D-01	
EPSY	-0.721166/D-03	SIGYY	0.870975D+02		
EPSZ	-0.119109D-03	SIGZZ	0.8336063D+01		
EPSXZ	-0.476125D-05	TAUXZ	-0.321518D+01	STRAIN ENERGY = 0.1625163D+01	
EPSYZ	-0.335311D-05	TAUYZ	-0.219666D+01		
EPSZY	-0.237889/D-05	TAUZY	0.122799D+00		
• ELEMENT (29) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (1) •					
EPSX	0.100000D-02	SIGXX	0.290933D+00	COORDINATES OF OUTPUT POINTS Y = 0.200000D+00 Z = 0.100000D-01	
EPSY	-0.721166/D-03	SIGYY	0.870959D+02		
EPSZ	-0.120615D-03	SIGZZ	-0.333279D+01		
EPSXZ	-0.369633D-05	TAUXZ	-0.109638D+02	STRAIN ENERGY = 0.1625093D+01	
EPSYZ	-0.369578D-05	TAUYZ	-0.121638D+02		
EPSZY	-0.237889/D-05	TAUZY	0.122560D+00		
• ELEMENT (29) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (7) •					
EPSX	0.100000D-02	SIGXX	0.290097D+00	COORDINATES OF OUTPUT POINTS Y = 0.200000D+00 Z = 0.100000D-01	
EPSY	-0.721166/D-03	SIGYY	0.171133D+02		
EPSZ	-0.120615D-03	SIGZZ	-0.166005D+02		
EPSXZ	-0.369719D-05	TAUXZ	-0.187150D+02	STRAIN ENERGY = 0.1625065D+01	
EPSYZ	-0.369601D-05	TAUYZ	-0.162909D+02		
EPSZY	-0.237889/D-05	TAUZY	0.122940D+00		
• ELEMENT (29) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (9) •					
EPSX	0.100000D-02	SIGXX	0.291129D+00	COORDINATES OF OUTPUT POINTS Y = 0.200000D+00 Z = 0.100000D-01	
EPSY	-0.721166/D-03	SIGYY	0.109585D+01		
EPSZ	-0.110763D-03	SIGZZ	0.835056D+01		
EPSXZ	-0.369667D-05	TAUXZ	-0.496918D+01	STRAIN ENERGY = 0.1627918D+01	
EPSYZ	-0.126733D-05	TAUYZ	-0.468295D+00		
EPSZY	-0.237889/D-05	TAUZY	-0.129671D+00		
• ELEMENT (29) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (3) •					
EPSX	0.100000D-02	SIGXX	0.292633D+00	COORDINATES OF OUTPUT POINTS Y = 0.200000D+00 Z = 0.100000D-01	
EPSY	-0.721166/D-03	SIGYY	0.103670D+01		
EPSZ	-0.120615D-03	SIGZZ	-0.328260D+01		
EPSXZ	-0.369695D-05	TAUXZ	-0.129007D+02	STRAIN ENERGY = 0.1627912D+01	
EPSYZ	-0.369671D-05	TAUYZ	-0.127949D+02		
EPSZY	-0.237889/D-05	TAUZY	-0.129751D+00		
• ELEMENT (29) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (8) •					
EPSX	0.100000D-02	SIGXX	0.292131D+00	COORDINATES OF OUTPUT POINTS Y = 0.200000D+00 Z = 0.100000D-01	
EPSY	-0.721166/D-03	SIGYY	0.286578D+02		
EPSZ	-0.120615D-03	SIGZZ	-0.169157D+02		
EPSXZ	-0.369580D-05	TAUXZ	0.301766D+02	STRAIN ENERGY = 0.1631032D+01	
EPSYZ	-0.369599D-05	TAUYZ	-0.660086D+02		
EPSZY	-0.237889/D-05	TAUZY	-0.129836D+00		

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• ELEMENT (29) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (5) •

STRESS
SIGXX 0.294376D+00
SIGYY -0.294376D+00
SIGZZ 0.316933D+00
TAUZY 0.101096D+00
TAUZX -0.566933D+00
TAUZY 0.566933D+00
STRAIN ENERGY 0.115677D+00

• ELEMENT (29) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (2) •

STRESS
SIGXX 0.211716D+00
SIGYY -0.116626D+00
SIGZZ -0.613013D+00
TAUZY 0.293066D+00
TAUZX -0.838035D+00
TAUZY 0.538035D+00
STRAIN ENERGY 0.116192D+00

• ELEMENT (30) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (5) •

STRESS
SIGXX 0.296871D+00
SIGYY -0.296871D+00
SIGZZ 0.283075D+00
TAUZY 0.126073D+00
TAUZX -0.168529D+00
TAUZY 0.581131D+00
STRAIN ENERGY 0.119687D+00

• ELEMENT (31) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (5) •

STRESS
SIGXX 0.298189D+00
SIGYY -0.298189D+00
SIGZZ 0.209731D+00
TAUZY 0.200783D+00
TAUZX -0.299731D+00
TAUZY 0.743786D+00
STRAIN ENERGY 0.129919D+00

• ELEMENT (32) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (5) •

STRESS
SIGXX 0.289133D+00
SIGYY 0.195690D+00
SIGZZ 0.893003D+00
TAUZY 0.112963D+00
TAUZX -0.462389D+00
TAUZY 0.114975D+00
STRAIN ENERGY 0.151106D+00

• ELEMENT (32) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (6) •

STRESS
SIGXX 0.111575D+00
SIGYY 0.380229D+00
SIGZZ 0.262103D+00
TAUZY 0.335076D+00
TAUZX -0.662632D+00
TAUZY 0.142872D+00
STRAIN ENERGY 0.143669D+00

• ELEMENT (32) STRESSES, STRAINS AND STRAIN ENERGIES PER UNIT VOLUME EVALUATED AT POINT (1) •

STRESS
SIGXX 0.100000D+00
SIGYY 0.240000D+00
SIGZZ 0.100000D+00
TAUZY 0.100000D+00
TAUZX 0.100000D+00
STRAIN ENERGY 0.143669D+00

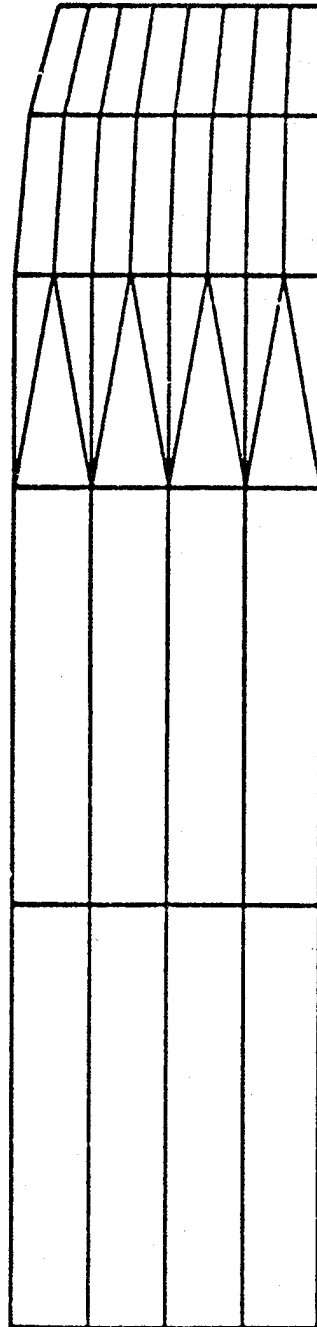
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ITEM 1	0.6/500270-01	SIGX	0.3600131003	Y	0.2900000000
ITEM 2	0.1/118570-01	SIGY	-0.3010013002	Z	0.1000000000
ITEM 3	0.2/150810-01	TAUX	0.1300078002		
ITEM 4	0.3/11510-01	TAUY	-0.0618175003		
ITEM 5	0.1/678010-05	TAUZ	0.1037680000		
STRAIN ENERGY = 0.1610411001					
COORDINATES OF OUTPUT POINTS					
ITEM 1	0.10000000-02	SIGX	0.3083530000	Y	0.2900000000
ITEM 2	0.6/200790-03	SIGY	0.3363817003	Z	0.1000000000
ITEM 3	0.18070670-03	TAUX	-0.0641063002		
ITEM 4	0.96499890-01	TAUY	0.0005010002		
ITEM 5	0.16785900-02	TAUZ	-0.8573719003		
ITEM 6	0.1/678010-05	TAUZ	0.1036640000		
STRAIN ENERGY = 0.2060800001					
COORDINATES OF OUTPUT POINTS					
ITEM 1	0.10000000-02	SIGX	0.2900000000	Y	0.2900000000
ITEM 2	0.6/200790-03	SIGY	0.3083530000	Z	0.1000000000
ITEM 3	0.18070670-03	TAUX	-0.0641063002		
ITEM 4	0.96499890-01	TAUY	0.0005010002		
ITEM 5	0.16785900-02	TAUZ	-0.8573719003		
ITEM 6	0.1/678010-05	TAUZ	0.1036640000		
STRAIN ENERGY = 0.1516091001					
COORDINATES OF OUTPUT POINTS					
ITEM 1	0.10000000-02	SIGX	0.1167199000	Y	0.2900000000
ITEM 2	0.6/200790-03	SIGY	0.3083530000	Z	0.1000000000
ITEM 3	0.18070670-03	TAUX	-0.0641063002		
ITEM 4	0.96499890-01	TAUY	0.0005010002		
ITEM 5	0.16785900-02	TAUZ	-0.8573719003		
ITEM 6	0.1/678010-05	TAUZ	0.1036640000		
STRAIN ENERGY = 0.1041171001					
COORDINATES OF OUTPUT POINTS					
ITEM 1	0.10000000-02	SIGX	0.3133998000	Y	0.2900000000
ITEM 2	0.6/200790-03	SIGY	0.3363817003	Z	0.1000000000
ITEM 3	0.18070670-03	TAUX	-0.0641063002		
ITEM 4	0.96499890-01	TAUY	0.0005010002		
ITEM 5	0.16785900-02	TAUZ	-0.8573719003		
ITEM 6	0.1/678010-05	TAUZ	0.1036640000		
STRAIN ENERGY = 0.1616197001					
COORDINATES OF OUTPUT POINTS					
ITEM 1	0.10000000-02	SIGX	0.3100798000	Y	0.2900000000
ITEM 2	0.6/200790-03	SIGY	0.3363817003	Z	0.1000000000
ITEM 3	0.18070670-03	TAUX	-0.0641063002		
ITEM 4	0.96499890-01	TAUY	0.0005010002		
ITEM 5	0.16785900-02	TAUZ	-0.8573719003		
ITEM 6	0.1/678010-05	TAUZ	0.1036640000		
STRAIN ENERGY = 0.2067780001					
COORDINATES OF OUTPUT POINTS					
ITEM 1	0.10000000-02	SIGX	0.2989951000	Y	0.2900000000
ITEM 2	0.6/200790-03	SIGY	0.3083530000	Z	0.1000000000
ITEM 3	0.18070670-03	TAUX	-0.0641063002		
ITEM 4	0.96499890-01	TAUY	0.0005010002		
ITEM 5	0.16785900-02	TAUZ	-0.8573719003		
ITEM 6	0.1/678010-05	TAUZ	0.1036640000		
STRAIN ENERGY = 0.1516091001					

5		10	15	24	33	42
4	1	9	5	10	23 21	32 29
			14	11	22 22	31 30
3	2	8	6	12 21	23	30 31
			13	14 20 24		29 32
2	3	7	7	15 19 25		28 33
			12	17 18 26		27 34
1	4	6	8	18 17 27		26 35
			11	20 16 28		25 36
						34

Example Problem, T300/5208, 8-Layer, (45.0,45.0,-45.0,-45.0)_s
 EPSX = 0.001, Hygrothermal Load = 0.000

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Example Problem, T300/5208, 8-Layer, (45.0, 45.0, -45.0, -45.0)_s
EPSX = 0.001, Hygrothermal Load = 0.000